| Name: <br> Enrolment No: | UPごS |  |  |
| :---: | :---: | :---: | :---: |
| Cours <br> Progr <br> Cours <br> Instru | UNIVERSITY OF PETROLEUM AND ENERGY STUDIES <br> Supplementary Examination, Dec. 2023 <br> Aircraft Structure-I <br> B. Tech ASE/ASE-AVE <br> Code: ASEG 3010 <br> ions: a) All questions are compulsory. <br> b) Assume any suitable value for the missing data <br> c) Q1-Q3 are TRUE/FALSE | : V hrs. rks: |  |
| SECTION A |  |  |  |
| S. No. |  | $\begin{gathered} \text { Mar } \\ \text { ks } \\ \hline \end{gathered}$ | CO |
| Q 1 | a) Two bars with same mass has always same strain energy under load <br> b) Within the proportional limit, strain energy varies linearly with load | 4 | CO1 |
| Q2. | c) The Change in slope between any two points on the elastic curve equals the area of the $\mathrm{M} / \mathrm{EI}$ diagram between both endpoints of the beam. <br> d) The method of the section can be applied if the section cuts four members. | 4 | CO4 |
| Q3. | a) Strain energy method to determine deflection applicable only for all elastic material <br> b) A column is free from both ends, if the column is heated uniformly across length then the column can buckle due to increase in temperature | 4 | CO 3 |
| Q4. | Compare the critical stresses using Euler's and Rankine's formulae for struts with slenderness ratios 50, Assume that both ends are hinged. $\mathrm{E}=200 \mathrm{GPa}$, Rankine's constant $=1 / 7500$, and $\sigma_{y}=300 \mathrm{MPa}$ | 4 | CO4 |
| Q5. | Two bars are subjected a load of 20 KN as shown in fig. below, determine the vertical If the radius of each members is 25 mm and $\mathrm{E}=200 \mathrm{GPa}$. The vertical deflection of point B is ..... | 4 | CO2 |


|  |  |  |  |
| :---: | :---: | :---: | :---: |
|  | SECTION B |  |  |
| Q 6 | For the propped cantiliver beam below determine the support reaction if $\mathrm{M}_{0}=10 \mathrm{KNm}$ and draw the shear force and bending moment. | 10 | CO2 |
| Q7. | Figure shows the cross sections of two aluminum alloy 2114-T6 bars that are used as compression members, each with effective length of L. Find (a) the wall thickness the hollow square bar so that the bars have the same cross-sectional area; (b) the critical load of each bar. Given: $\mathrm{L}=3 \mathrm{~m}$ and $\mathrm{E}=72 \mathrm{GPa}$. | 10 | CO3 |
| Q8. | An element in plane stress on the fuselage of an airplane (figure part a) is subjected to compressive stresses with a magnitude of 42 MPa in the horizontal direction and tensile stresses with a magnitude of 9.5 MPa in the vertical direction (see figure part b). Also, shear stresses with a magnitude of 15.5 MPa act in the directions shown, Determine the stresses acting on an element oriented at a clockwise angle of 400 from the horizontal. Show these stresses on a sketch of an element oriented at this angle. | 10 | CO 3 |


|  |  |  |  |
| :---: | :---: | :---: | :---: |
| Q9 | The load on a bolt consists of an axial pull of I0 kN together with a transverse shear force of 6 kN . Calculate the diameter of the bolt according to <br> (a) Maximum principal strain theory <br> (b) Shear strain energy theory. <br> Take factor of safety as 2 , given yield strength of the material $=310 \mathrm{~N} / \mathrm{mm} 2$ and poisson's ratio $=0.27$ | 10 | CO4 |
| SECTION-C |  |  |  |
| Q10 | Determine the vertical deflection of point C due to the applied load | 20 | CO 2 |
| Q11. | Analyze the continuous beam as shown below using and draw the SF and BM diagrams. <br> OR <br> Analyse the beam and draw the shear force and bending moment diagram. | 20 | CO1 |



