Name:

**Enrolment No:** 



## UNIVERSITY OF PETROLEUM AND ENERGY STUDIES End Semester Examination, December 2019

Course: Digital Avionics Program: B. Tech ASE+AVE Course Code: AVEG 433 Semester: VII Time 03 hrs. Max. Marks: 100

Instructions: Make use of *sketches/plots* to elaborate your answer. Brief and to the point, answers are expected. The Question paper has three sections: Section A, B and C, Section B and C having internal choices. SECTION A

SECTION A		
S. No. Questions	Marks	CO
Q 1 Write shorts note on fiber optic Data buses <b>DOD-STD 1773</b>	4	CO3
Q 2 What are the components of feedback control system and types of feedback (FB) is employed in control systems. Explain the Effects of FB in Automatic Flight Control Systems ( <b>AFCS</b> ).	4	CO2
Q 3 What are the major factors consider designing the Helmet Mounted Display ( <b>HMD</b> ) of Fighter Aircraft.	4	<b>CO4</b>
Q 4 Explain the various role in civil and military aircraft.	4	CO1
Q 5Discuss the Dead-Reckoning ( <b>DR</b> ) Navigation system with suitable examples	4	CO5
SECTION B		
For the following clock pulse explain the Manchester Bi-Phase coding and find the Data. Clock Manchester (as per G.E. Thomas) Manchester (as per IEEE 802.3) Also, state the Data buses of MIL-STD 1533B Military Aircraft protocols.	10	CO3

Q 7	Write the MATLAB programming for vanguard missile control system, amplifier gain $S_{(amp)} = 10$ $\theta_i  \varepsilon  S_{(amp)}  \varepsilon_{\sigma}  13.3(\varepsilon + 2.26)  \delta_i  Servo  \delta  Missile  \theta_o$ Lead networkVanguard control system (rigid missile)TF (servo) $= \frac{2750}{(S^2 + 84S + 2750)};$ TF (Missile) $= \frac{-7.21}{(S+1.6)(s-1.48)}$	10	CO1
Q 8	If (set vo) $-(s^2+84s+2750)$ ,       If (unsult) $-(s+1.6)(s-1.48)$ Find the following conversion:       a) (132)10 to Binary         b) (73.75)10 to Octal       c) (137.21)8 to Decimal         d) (C3A6)16 to Binary       e) (82.25)10 to Hex equivalent	10	CO3
Q 9	<ul> <li>a) A transmitter uses a single error-correcting code for the message using even parity. The message received at the receiving end is 1110101. Check and correct the error.</li> <li>b) Find the required effective focal length <i>F</i>, Head up display (HUD) for civil aircraft TFOV of 20<sup>0</sup> and a CRT diameter of 50 mm.</li> <li>(Or)</li> <li>a) The response of a servomechanism is c(t) = 1+ 0.2 e<sup>-60t</sup> - 1.2 e<sup>-10t</sup> when subject to a unit step input. Obtain an expression for the system</li> <li>b) The following equation S<sup>4</sup> +0.811S<sup>3</sup> + 1.32S<sup>2</sup> +0.0102S +0.00695 = 0. Find the damping Ratio and undamped natural frequency for <ul> <li>i) Phugoid oscillation</li> <li>ii) Short-Period oscillation</li> </ul> </li> </ul>	10	CO4

	SECTION C		
Q 10	How GPS system works? A satellite transmit a signal at the nominal GPS time (by its clock) of $t_{sv}$ , However, the clock corrections broadcast in the data stream indicates		
Q IO	corrections $\Delta t_{sv}$ , to be added to the satellite clock time. The signal is received by the	20	
	user at time $\Delta t_{u_{i}}$	20	CO5
	By the user clock, which has got an error indicated by $t_{bias}$ , Write the range equations for the satellite which takes these into consideration. Show by a sketch how these factors affect the measured delay.		
	Convert the block diagram to signal flow graph and determine the transfer function using mason's Gain formula		
Q 11	$\begin{array}{c} R(s) + & V_{1}(s) \\ \hline \\ \\ \\ \\ \\ \\ \\ \\ \\ \\ \\ \\ \\ \\ \\ \\ \\ $		
	(Or)		
	Reduce the block diagram shown in figure to a single block $\frac{C(s)}{R(s)}$	20	CO2
	$R(s) + G_1(s) + G_5(s) + G_6(s)$		
	$G_2(s)$		
	$\begin{array}{c} -2 \langle s \rangle \\ + \\ & & \\ & $		