


Name: Enrolment No:			
UPES End Semester Examination, May 2025			
Course: Space Communication Program: B.Tech Aerospace Engg. Course Code: ASEG3036P		Semester: VI Time : 03 hrs. Max. Marks: 100	
Instructions: All questions are compulsory. Neat diagrams and real-world examples are encouraged.			
SECTION A (5Qx4M=20Marks)			
S. No.		Marks	CO
Q 1	What is the significance of satellite constellation configuration? Explain with suitable examples.	4	CO1
Q 2	Define transient temperature variation in orbit. How does it affect satellite operations?	4	CO3
Q 3	Differentiate between telemetry and telecommand systems.	4	CO1
Q 4	Explain the impact of microgravity and thermal cycling on spacecraft structure.	4	CO3
Q 5	What role do sun sensors and magnetometers play in satellite attitude control?	4	CO3
SECTION B (4Qx10M= 40 Marks)			
Q 6	Analyze the design trade-offs between electric and chemical propulsion systems for Earth-orbiting satellites.	10	CO4
Q 7	Evaluate the subsystem integration strategy adopted in the James Webb Space Telescope. Focus on structural, thermal, and TT&C systems.	10	CO2
Q 8	Discuss the key features and advancements of modern ground control systems. How do they support autonomous spacecraft operations? OR Compare passive and active thermal control techniques with examples from planetary and deep-space missions.	10	CO3 Or CO4
Q 9	Describe the typical RF link budget considerations in TT&C systems for LEO satellites.	10	CO3
SECTION-C (2Qx20M=40 Marks)			
Q 10	a) Derive and formulate the transfer function of a basic RC low-pass filter and describe how component values influence its cutoff frequency.	10+10= 20	CO2

	<p>b) An RC low-pass filter is built using a resistor $R = 1 \text{ k}\Omega$ and a capacitor $C = 0.1 \text{ }\mu\text{F}$. (i) Calculate the cutoff frequency (f_c) of the filter. (ii) If a sinusoidal signal of frequency 1 kHz is applied, will it be attenuated significantly?</p> <p style="text-align: center;">OR</p> <p>Develop a thermal protection strategy for a satellite designed for operations in geostationary orbit. Highlight control techniques and integration constraints.</p>		
Q 11	Evaluate the telemetry and command strategy used in the Perseverance Mars Rover. Discuss its robustness, redundancy, and communication architecture.	20	CO4